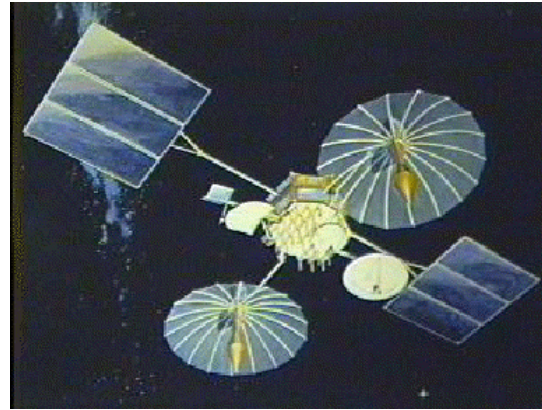
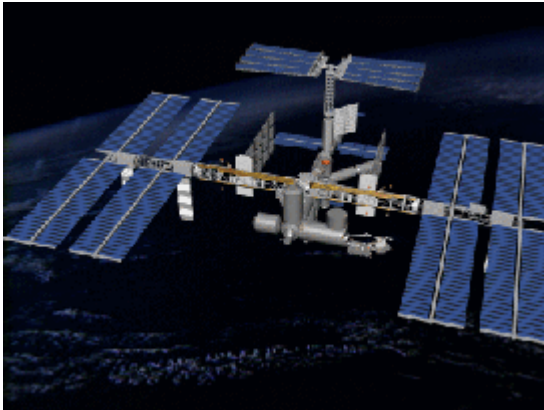


INTERNATIONAL SPACE STATION

COMMUNICATIONS, TRACKING, AND DATA ACQUISITION REQUIREMENTS AND OPERATIONS CONCEPTS



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ABSTRACT

The main goal of the Space Network (SN) is to provide multiple crewed and robotic flight programs with high-quality, very reliable, global Communications, Tracking, and Data Acquisition (CT&DA) services. The most ambitious and complex of these programs is the International Space Station (ISS).

The ISS and associated re-supply missions referred to as Visiting Vehicles, require multiple space-to-ground Radio Frequency (RF) links as well as a global network of ground communication paths to join vehicle and payload operations with terrestrial-based controllers and scientists.

To understand the requirements the ISS and Visiting Vehicles have placed on the SN, it is convenient to organize them into three primary components: the Core Vehicle, Science, and Visiting Vehicle CT&DA services. These requirements are in some cases reasonably firm and complete, while others are evolutionary. Coordination of these services is on-going and involves several ISS teams and International Partner (IP) forums.

This paper provides a brief discussion of the requirements, both firm and evolving, and the SN planning and technical approaches to meeting the ISS's future service needs. Information on ground systems communication and data transport configurations are not addressed herein; however, this information can be provided upon request.

SECTION 1. INTRODUCTION

The current SN architecture has progressed from a basic single satellite, main-frame based ground station to a multiple satellite, dual ground station system in response to expanded customer service support requirements, while taking full advantage of the evolution of computer and electronic technology. The ISS era is stimulating the application of continually improving technologies coupled with innovative SN operations techniques to ensure continued, high-quality customer support. To address the progression of enhanced and expanded SN capabilities, this paper is organized to first outline the ISS and Visiting Vehicle requirements, followed by information on the SN's evolution to today's configuration and operations concepts, and then concluding with the Tracking and Data Relay Satellite System (TDRSS) service characteristics and capabilities planned to be in place to meet all of the SN customer's needs during the ISS era.

SECTION 2. ISS CT&DA REQUIREMENTS

Clear, unambiguous agreements between SN customers and service providers represent one of the key planning tools essential to enabling the manager's ability to seek and obtain reconciliation between flight program CT&DA needs and support system capabilities. For the ISS, these agreements (or requirements) are quite complex and long-term, involving large expenditures and representing several years of coordination and cost-benefit trades.

ISS requirements are presented in two tables. Table 1A contains documented (firm) requirements, divided into Core Vehicle, Telescience, and Visiting Vehicle CT&DA services. Table 1B is similarly structured and includes evolving requirements designed to accommodate the enhanced video and science needs of the ISS customers and visiting re-supply vehicles in the operations and utilization phase. Also included, in Table 1C and Figure A, are the known and anticipated radio systems operating as part of the overall ISS clustered in the S-band frequency spectrum.

Table 1A. Documented ISS/TDRSS Requirements

Date	FLT	Event	Requirement
International Space Station Communication			
July 1998	2A	Node 1/Early Communications (ECOMM)	S-band U/L: 6/128 kbps D/L: 20.48/128 kbps
March 1999	4A	S-band Low Data Rate (LDR) Activation	S-band U/L: 6 kbps D/L: 12 kbps
May 1999	5A	S-band High Data Rate (HDR) Activation	S-band U/L: 72 kbps D/L: 192 kbps
June 1999	6A	Ku-band Activation	Ku-band D/L: 50 Mbps
January 2000	UF-1	Communications Outage Recorder	–
Telescience Communication			
June 1999	6A	Lab Racks Space to Space – No FWD Link Supported – Video Baseband Signal Processor (VBSP) up to 4 channels of video	Ku-band D/L: 50 Mbps
June 2002	UF-5	Ku-band Upgrade – Ku-band FWD (RCVR & DEMUX) – Multi-Channel FWD Compressed Video & sync Audio – Multi-Channel Compressed Video FWD and RTN sync Audio	Ku-band D/L: 75 Mbps Ku-band U/L: 3-10 Mbps
October 2003	UF-7	Ku-band Upgrade	Ku-band D/L: 150 Mbps
Visiting Vehicle Communication			
–	–	Space Shuttle	S-band U/L: 72 kbps D/L: 192 kbps Ku-band U/L: 216 kbps D/L: 50 Mbps or Video

Table 1B. Evolving ISS/TDRSS Requirements

Date	FLT	Event	Requirement
International Space Station Communication			
July 1998	LV	Russian Service Module (SM)	S-band & Ku-band
December 1998	2A.1	Interim Control Module (ICM)	S-band U/L: 1 kbps D/L: 1-8 kbps
June 1999	6A	Ku-band Activation	Ku-band U/L: 3 Mbps
Telescience Communication			
Future (?)	TBD	Wideband Ka-band (Enhanced K-band Plan Phase III) – Expanded Bandwidth – Increased Coverage – High Speed Data Router FWD/RTN – Support High Definition TV (HDTV)	Ka-band D/L: 300 Mbps U/L: 25 Mbps
Visiting Vehicle Communication			
January 2001	LV	Inspector	S-band D/L: 1 kbps
March 2002	LV	Automated Transfer Vehicle (ATV)	S-band U/L: 2 kbps D/L: 8 kbps
October 2002	LV	H-II Transfer Vehicle (HTV)	S-band U/L: 250-500 kbps D/L: 2-4 kbps
March 2003	18A	Crew Return Vehicle (CRV)	S-band D/L: 512 kbps
Future	TBD	Spartan-400	S-band U/L: 2 kbps D/L: 2 Mbps
Future (2010)	–	Reusable Launch Vehicle (RLV)	S-band
Future	LV	Soyuz	S-band

Table 1C. S-band Frequencies

2287.5/2106.4 MHz (MA/SSA)	2217.5/2041.9 MHz (SSA)	To Be Determined
ECOMM	Assembly Contingency Subsystem (ACS)	Spartan-400
ATV	Space Shuttle	Inspector
HTV	CRV	RLV
Space Shuttle		SM
		ICM
		Soyuz

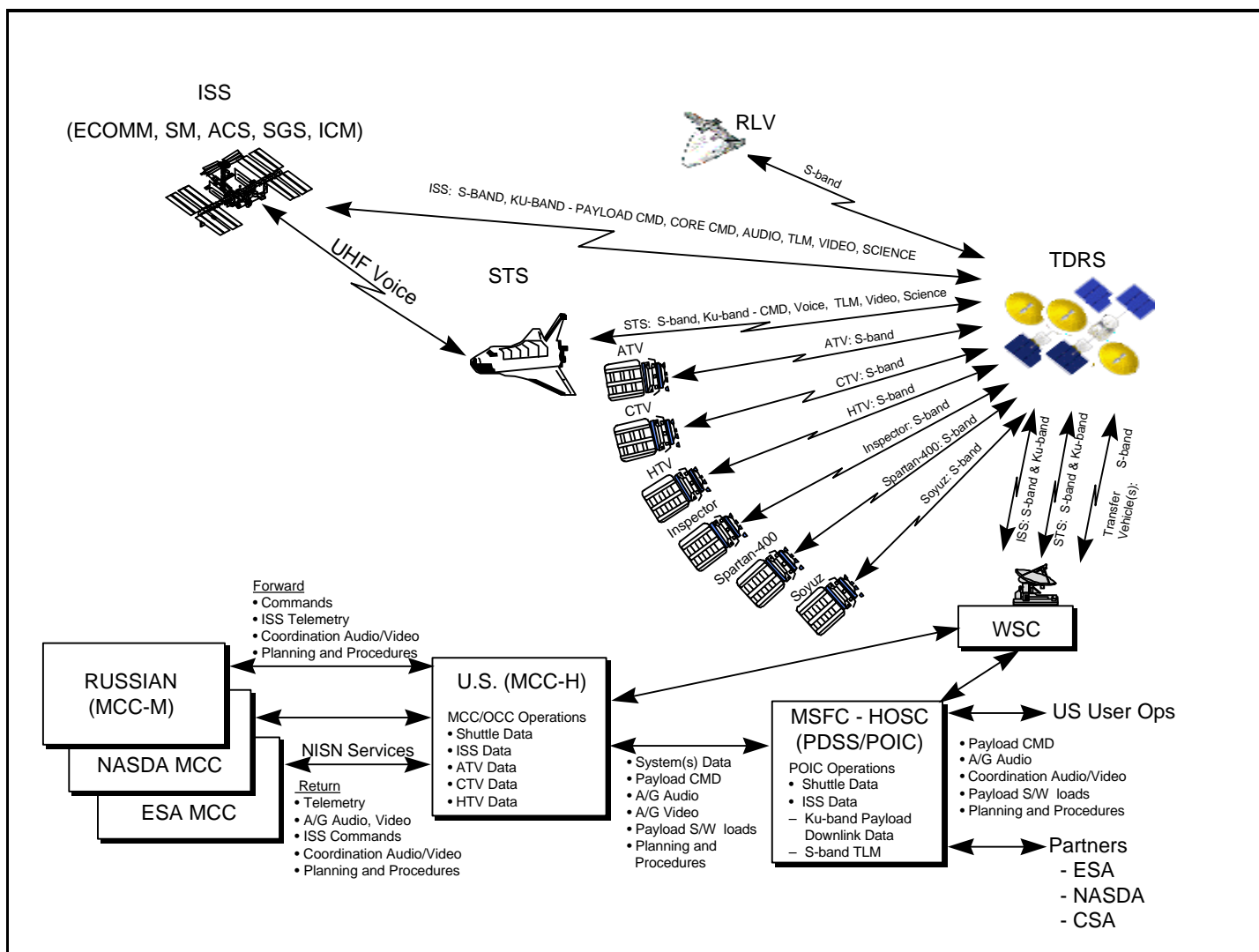


Figure A. ISS/TDRSS Links

SECTION 3. SN CONFIGURATION AND OPERATIONS CONCEPTS

The evolution of the SN from a single satellite, single ground station configuration to today's multiple satellite and ground stations configuration (Figure B) was accomplished because the anticipated customer utilization model supported the need for significant increases in supportable services. Another upgrade objective was to implement an architecture consistent with technological advances in Radio Frequency devices and data handling/management systems. The services provided by the early configuration are no different in the current system; however, there will be improved support capabilities included in the follow-on Tracking and Data Relay Satellites (TDRSs) H, I, and J implementation as discussed in Section 4.

The SN operations concept is driven by electronically-conveyed customer requests for specific services for determined times, using pre-established configurations and Radio Frequency (RF) parameters. The services provided include forward and return links with tracking if requested.

Forward data relay services are provided for S-band Single Access (SSA), S-band Multiple Access (SMA) or Ku-band Single Access (KSA) scheduled supports and are only handled as real-time activities via the telecommunications path between the flight project control center and the user spacecraft.

Return data relay services are also provided using the SSA, SMA, or KSA scheduled supports handled in real-time, or with data recorded and played-back for either rate buffering, line outage protection, or to meet unique data needs of the receiving control center. No provisions are available for processing customer data except for time tagging and blocking into a format suitable for commercial data transport systems.

The SN provides one- or two-way tracking data for customer spacecraft orbit determination in the form of range and range rate measurements. Range measurements require the return service to have a Pseudo-random Noise (PN) range code which is coherent with the forward service range channel PN code. Two-way Doppler measurements require carrier frequency coherency between the forward and return services. PN ranging and Doppler measurements can be provided in conjunction with SSA, SMA, or KSA services.

Detailed information concerning the functions and operational use of the SN are available in the *Space Network User's Guide*, 530-SNUG, and other technical documents available via the Internet/World Wide Web (WWW) at Uniform Resource Locator (URL) <http://tip.gsfc.nasa.gov/project/530other.htm>.

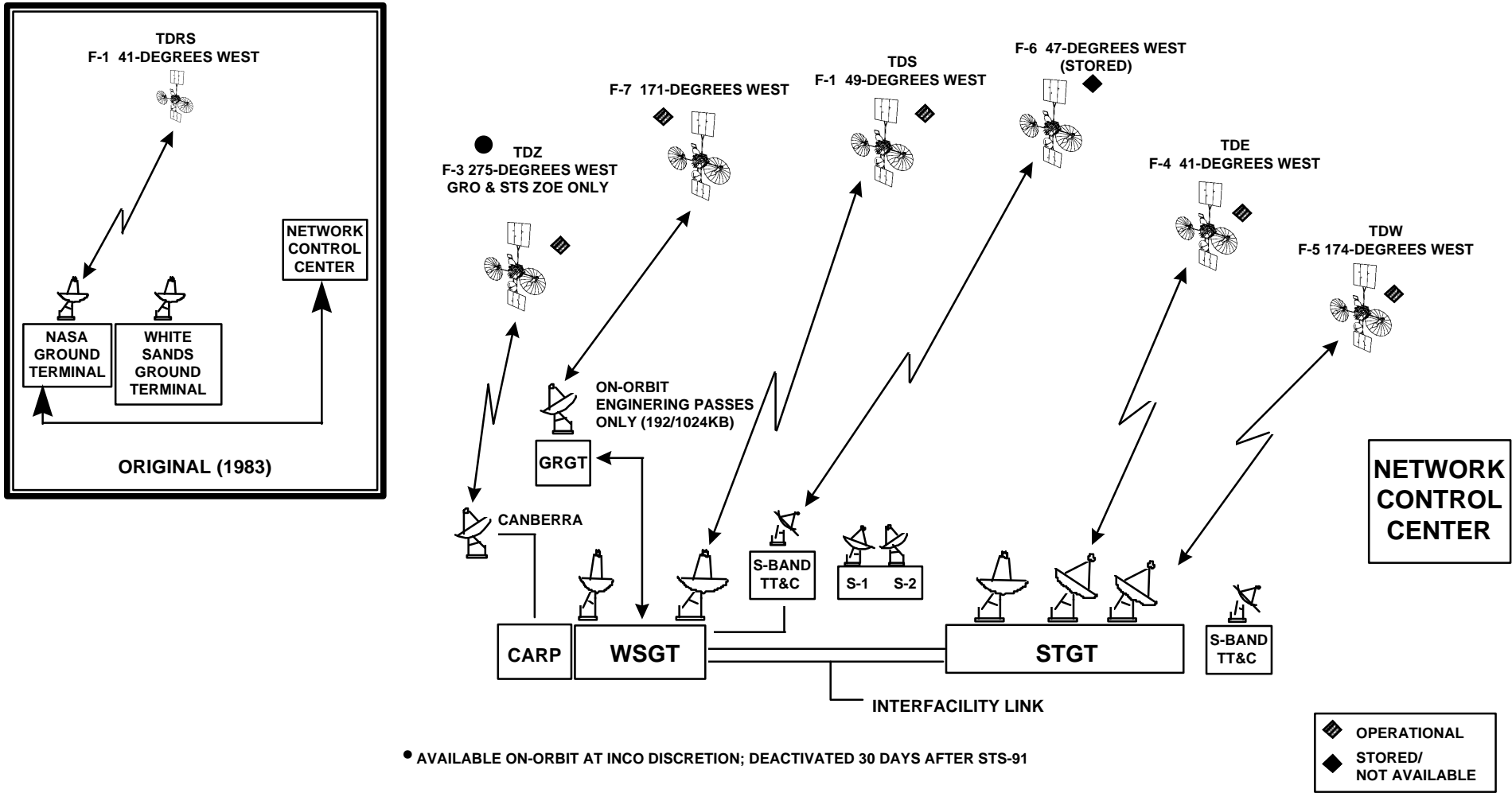


Figure B. Space Network Configuration - Original (1983) and Present (1998)

SECTION 4. TDRSS SERVICES

Future Capabilities: The next generation of Tracking and Data Relay Satellites (TDRSs), termed H, I, and J, and their associated changes to NASA's White Sands Complex (WSC) are being developed for NASA by the Hughes Space and Communications Group (HSC) to provide SN customer support into the 21st century. Six of the current fleet of TDRSs, termed A through G, are on-orbit and providing scheduled services almost continuously as either prime or spare resources. This section describes the user support services provided by the SN and contrasts the new H, I, and J capabilities with those of the A through G spacecraft. Table 4A summarizes these services as provided by the SN.

Table 4A. WSC Capabilities Comparison ⁽⁴⁾

Service			WSC TDRS A-G Capabilities	WSC TDRS H, I, J Capabilities
Single Access	S-band	Forward	300 kbps	300 kbps
		Return	6 Mbps	6 Mbps
	Ku-band	Forward	25 Mbps	25 Mbps
		Return	300 Mbps	300 Mbps
	Ka-band	Forward	————	25 Mbps
		Return	————	300 Mbps/800 Mbps ⁽¹⁾
Number of Links			2 SSA/TDRS; 12 SSA/WSC	2 SSA/TDRS; 12 SSA/WSC
			2 KuSA/TDRS; 12 KuSA/WSC	2 KuSA/TDRS; 12 KuSA/WSC
				2 KaSA/TDRS; 8 KaSA/WSC ⁽²⁾
Multiple Access		Forward	1/TDRS @ 10 kbps 4/WSC	2/TDRS ⁽³⁾ @ 300 kbps 4/WSC
		Return	5/TDRS @ 100 kbps 20/WSC	5/TDRS @ 3 Mbps 20/WSC
User Tracking			Range, 1- & 2-way Doppler	Range, 1- & 2-way Doppler (No Ka-band Tracking)
(1) Spacecraft Only; (2) Ku or Ka; (3) 1/TDRS at WSC; (4) For User Data Configurations, refer to the <i>Space Network User's Guide</i> (530-SNUG)				

Support: The TDRSs support the SN missions with a real-time, bent-pipe telecommunications concept in which the TDRS acts as a relay between the user spacecraft and WSC. The WSC provides RF to base-band signal processing, data distribution, and user tracking data processing. TDRS services across the fleet include S-band Multiple Access (MA) and S-/Ku-band Single Access (SA) using 4.9-meter steerable antennae. For the H, I, and J spacecraft, Ka-band services are also available. Data rates for the services described in the following paragraphs are shown in Table 4A. Effective Isotropic Radiated Power (EIRP) and Gain/Temperature (G/T) values applicable to ISS and Visiting Vehicles as well as other near-earth orbiters are provided in Table 4B. The frequency bands described in the following paragraphs are for the TDRS to/from user links.

S-band: S-band SA Forward (SSAF) service frequencies are 2030 MHz to 2113.5 MHz. Return (SSAR) service frequencies are 2200 MHz to 2300 MHz. SSAR BPSK, QPSK, SQPSK, and SQPN modulation with Rate 1/2 encoding is supported by the SN with right- or left-hand circular polarization.

Table 4B. TDRS H,I,J Forward and Return Link Capabilities

TDRS		Forward Link EIRP (dBw)			
H, I, J (Specification Values)		SSA	KuSA	MA	KaSA
		36.3 to 48.5 in 1.0 dB steps	39.0 to 49.0 in 1.0 dB steps	34.0 to 42.0 in 1.0 dB steps	59.5 Prog. Track 63.0 Autotrack
A-G (Current Operating Values)		43.6 Normal 46.1 High	46.5 Normal 48.5 High	34.0 Minimum	–
TDRS		Return Link G/T (dB/K)			
H, I, J (Specification Values)		SSA	KuSA	MA	KaSA
		8.5	18.4 Prog. Track 24.4 Autotrack	4.5	23.0 Prog. Track 26.5 Autotrack
A-G		8.5	23.0	-14.7 per Element Max. 30 Elements	–

Ku-band: The Ku-band Forward (KSAF) service frequency is 13775 MHz. The return frequency is 15003.4 MHz. Modulation schemes and polarization options are the same as the SSAR services with or without Rate 1/2 encoding.

Ka-band: User services supported via the TDRS H, I, and J spacecraft may make use of a new TDRSS service in the Ka-band frequency spectrum. Forward frequencies are tunable in 5-MHz steps, from 22555 MHz to 23545 MHz. Return frequencies are tunable in 25-MHz steps, from 25253.4 MHz to 27478.4 MHz. Polarization is selectable left- or right-hand circular, and modulation schemes are identical to the current TDRS A-G spacecraft for Ku-band.

Note

Simultaneous SA S-band and either Ku-band or Ka-band can be provided via each spacecraft SA antenna concurrent with MA services.

Multiple Access Services: MA services using the TDRS A-G spacecraft are provided through phased array antennas at S-band with a forward frequency of 2106.4 MHz and a return frequency of 2287.5 MHz. Beamforming for up to five simultaneous return services is performed at the WSC. Return service modulation is SQPN with Rate 1/2 encoding. Beamforming for forward services is performed on the TDRS spacecraft. Left-hand circular polarization is required.

MA services on the TDRS H, I, and J spacecraft use separate forward and return phased array antennae with on-board beamforming to provide up to five return and two forward services. Currently the WSC can only provide one forward service per TDRS spacecraft. As delineated in Table 4A, data rates available for MA user services are considerably higher than current TDRS spacecraft. Modulation schemes are identical to the SSAR services. Frequencies and polarization are the same as MA services on TDRS A-G.

Data Interleaving: Periodic convolutional deinterleaving and decoding is provided in the WSC for S-band services with data rates greater than 300 kbps.

Tracking Services: PN code ranging and Doppler tracking services are available for all TDRSS support except Ka-band. TDRSS auto-tracking of user spacecraft is offered when operating at both Ku-band and Ka-band.

ACRONYMS AND ABBREVIATIONS

<u>Term</u>	<u>Definition</u>
A-G	Air to Ground
ACS	Assemble Contingency Subsystem
ATV	Automated Transfer Vehicle
BPSK	Biphase (or Binary Phase) Shift Keyed (or Keying)
CRV	Crew Return Vehicle
CT&DA	Communications, Tracking, and Data Acquisition
dB	Decibel
dBW	Decibel Referred to 1 Watts
DEMUX	Demultiplexer
D/L	Downlink
ECOMM	Early Communications
EIRP	Effective Isotropic Radiated Power
FLT	Flight
FWD	Forward
G/T	Gain/Temperature
HDR	High Data Rate
HDTV	High Definition Television
HSC	Hughes Space and Communications Group
HTV	H-II Transfer Vehicle
ICM	Interim Control Module
IP	International Partner
ISS	International Space Station .
KaSA	Ka-band Single Access
kbps	Kilobits per Second
KSA	Ku-band Single Access
KSAF	Ku-band Forward
KuSA	Ku-band Single Access
LDR	Low Data Rate
LV	Launch Vehicle
MA	Multiple Access
Mbps	Megabits per Second
PN	Pseudo-random Noise
QPSK	Quadrphase Shift Keyed (or Keying)
RCVR	Receiver
RF	Radio Frequency
RLV	Reusable Launch Vehicle
SA	Single Access
SM	Service Module
SMA	S-band Multiple Access
SN	Space Network
SQPN	Staggered Quadrphase Pseudorandom Noise
SQPSK	Staggered QPSK
SSA	S-band Single Access
SSAF	S-band SA Forward
SSAR	S-band SA Return
TBD	To Be Determined
TDRS	Tracking and Data Relay Satellite
TDRSS	Tracking and Data Relay Satellite System
UF	Utility Flight
U/L	Uplink
URL	Uniform Resource Locator
VBSP	Video Baseband Signal Processor
WSC	White Sands Complex
WWW	World Wide Web